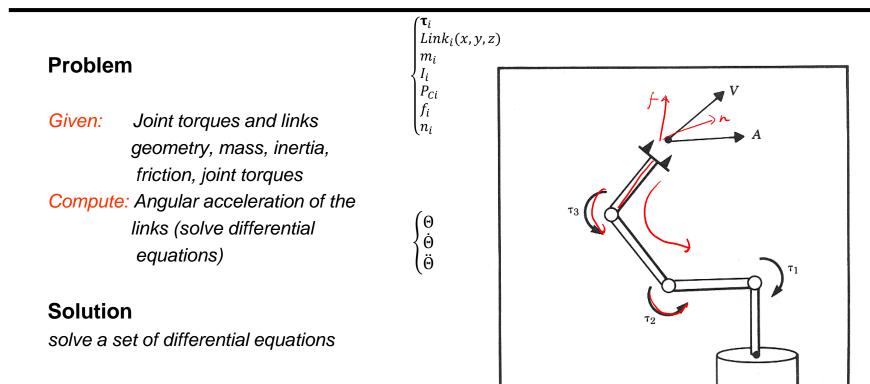


Manipulator Dynamics 2





Forward Dynamics



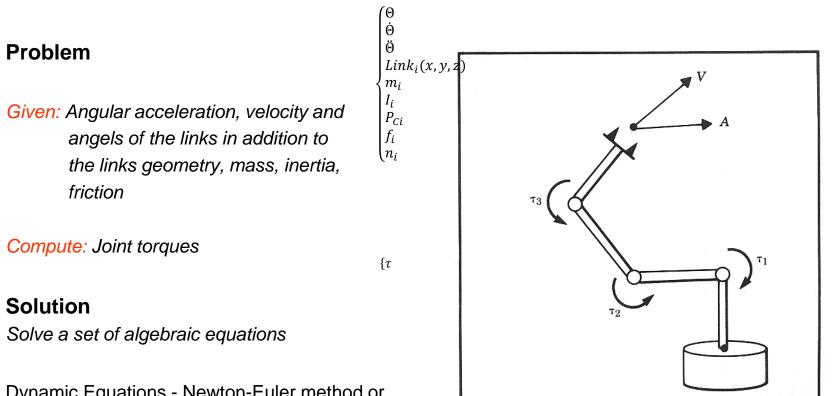
Dynamic Equations - Newton-Euler method or Lagrangian Dynamics

$$\mathbf{\tau} = M(\Theta)\ddot{\Theta} + V(\Theta,\dot{\Theta}) + G(\Theta) + F(\Theta,\dot{\Theta})$$

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Inverse Dynamics



Dynamic Equations - Newton-Euler method or Lagrangian Dynamics

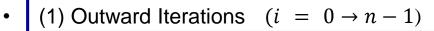
$$\mathbf{\tau} = M(\Theta)\ddot{\Theta} + V(\Theta,\dot{\Theta}) + G(\Theta) + F(\Theta,\dot{\Theta})$$







Iterative Newton Euler Equations Steps of the Algorithm



- Starting With velocities and accelerations of the base

 ${}^{0}\omega_{0} = 0$, ${}^{\dot{0}}\omega_{0} = 0$, ${}^{0}\nu_{0} = 0$, ${}^{\dot{0}}\nu_{0} = +g\hat{z}$

 Calculate velocities accelerations, along with forces and torques (at the CM)

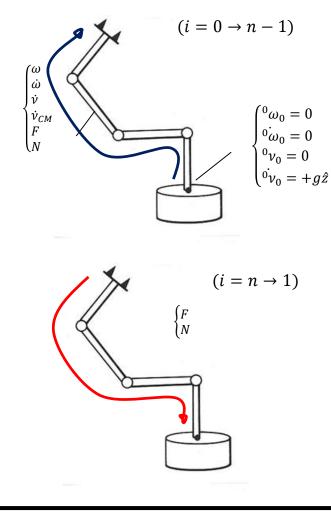
 $\omega, \dot{\omega}, \dot{\nu}, \dot{\nu}_{CM}, F, N$

- (2) Inward Iteration $(i = n \rightarrow 1)$
 - Starting with forces and torques (at the CM)

F, N

Calculate forces and torques at the joints

f,n







Iterative Newton-Euler Equations - Solution Procedure Phase 1: Outward Iteration

Outward Iteration: $i: 0 \rightarrow 5$ Calculate the link velocities and accelerations iteratively from the robot's base to the end effector . ${}^{i+1}\omega_{i+1} = {}^{i+1}R^{i}\omega_{i} + \dot{\theta}_{i+1}{}^{i+1}\hat{Z}_{i+1}$ ${}^{i+1}\dot{\omega}_{i+1} = {}^{i+1}_{i}R^{i}\dot{\omega}_{i} + {}^{i+1}_{i}R^{i}\omega_{i} \times \dot{\theta}_{i+1}{}^{i+1}\hat{Z}_{i+1} + \ddot{\theta}_{i+1}{}^{i+1}\hat{Z}_{i+1}$ ${}^{i+1}\dot{v}_{i+1} = {}^{i+1}_{i}R({}^{i}\omega_{i} \times {}^{i}P_{i+1} + {}^{i}\omega_{i} \times ({}^{i}\omega_{i} \times {}^{i}P_{i+1}) + {}^{i}\dot{v}_{i})$ ${}^{i+1}\dot{v}_{C_{i+1}} = {}^{i+1}\dot{\omega}_{i+1} \times {}^{i+1}P_{C_{i+1}} + {}^{i+1}\omega_{i+1} \times ({}^{i+1}\omega_{i+1} \times {}^{i+1}P_{C_{i+1}}) + {}^{i+1}\dot{v}_{i+1}$ Calculate the force and torques applied on the CM of each link using the Newton and Euler equations ٠ $^{i+1}F_{i+1} = m_{i+1}{}^{i+1}\dot{v}_{C_{i+1}}$ $^{i+1}N_{i+1} = {}^{C}{}^{i+1}I_{i+1}{}^{i+1}\dot{\omega}_{i+1} + {}^{i+1}\omega_{i+1} \times {}^{C}{}^{i+1}I_{i+1}{}^{i+1}\omega_{i+1}$

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Iterative Newton-Euler Equations - Solution Procedure Phase 2: Inward Iteration

Inward Iteration: $i: 6 \rightarrow 1$

• Use the forces and torques generated at the joints starting with forces and torques generating by interacting with the environment (that is, tools, work stations, parts etc.) at the end effector all the way the robot's base.

 ${}^{i}f_{i} = {}^{i}_{i+1}R^{i+1}f_{i+1} + {}^{i}F_{i}$

$${}^{i}n_{i} = {}^{i}N_{i} + {}^{i}_{i+1}R^{i+1}n_{i+1} + {}^{i}P_{C_{i}} \times {}^{i}F_{i} + {}^{i}P_{i+1} \times {}^{i}_{i+1}R^{i+1}f_{i+1}$$

 $\tau_i = {}^{i+1} n^T {}_{i+1} \; {}^{i} \widehat{Z}_i$





Manipulator Dynamics – Newton Euler Equations

The Inertia Tensor (Moment of Inertia)

сI

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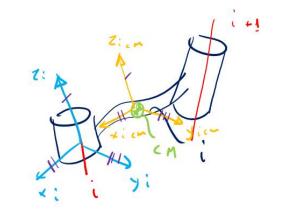


Dynamics - Newton-Euler Equations

- To solve the Newton and Euler equations, we'll need to develop mathematical terms for:
 - \dot{v}_c The linear acceleration of the center of mass
 - $\dot{\omega}$ The angular acceleration
 - ^cI The Inertia tensor (moment of inertia)
 - F The sum of all the forces applied on the center of mass
 - N The sum of all the moments applied on the center of mass

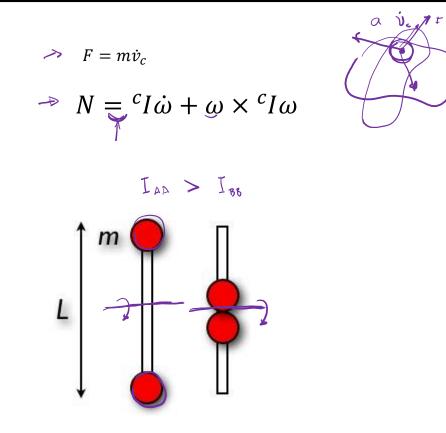
$$F = m\dot{v}_c$$

 $N = {}^{c}I\dot{\omega} + \omega \times {}^{c}I\omega$









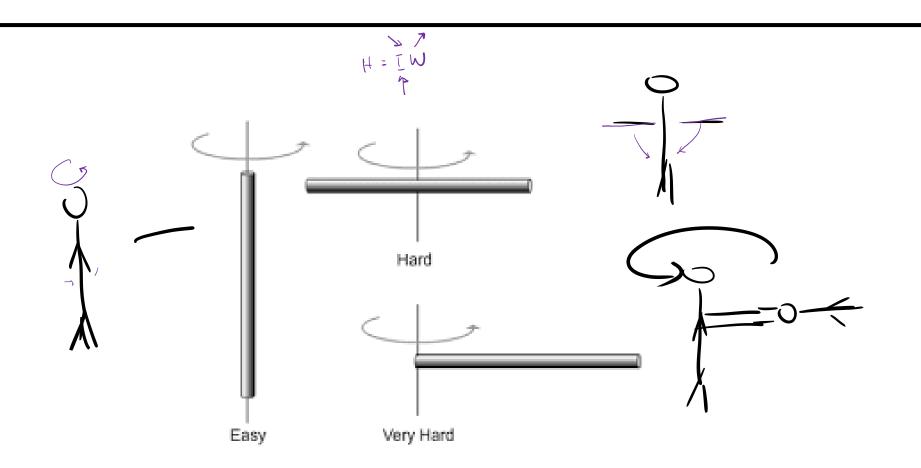






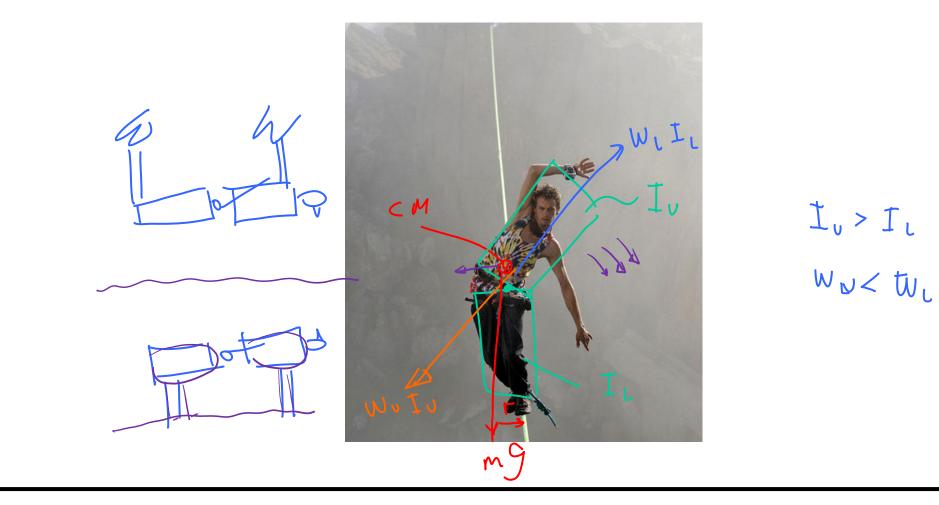












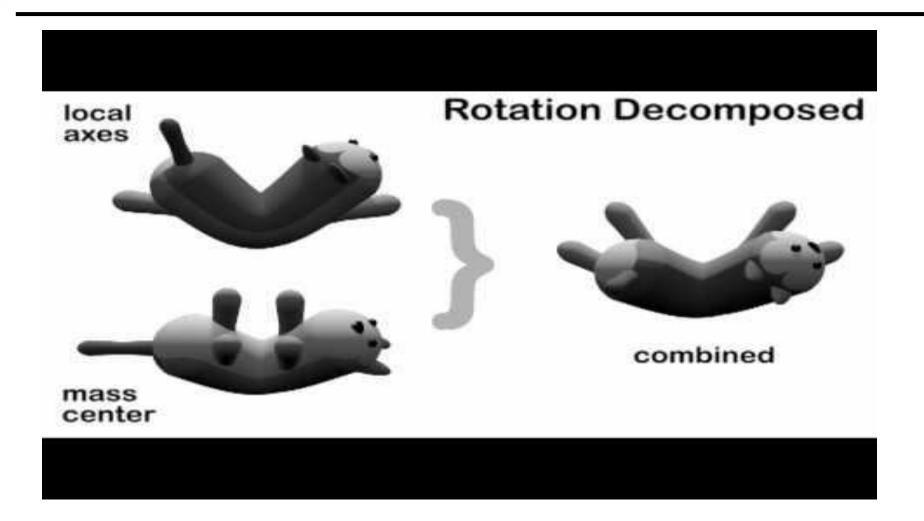
















<u>https://www.youtube.com/watch?v=9SaShn8OkJl</u>



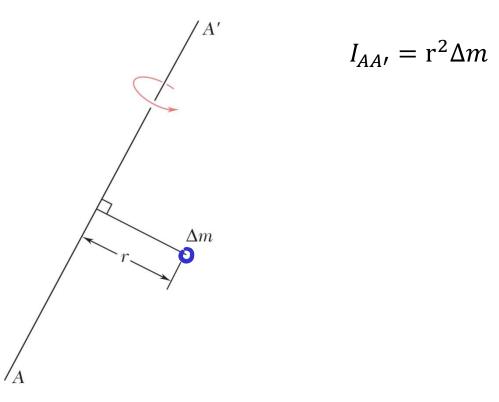








Moment of Inertia – Particle – WRT Axis

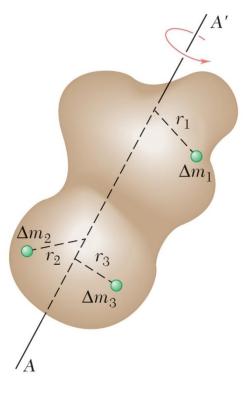






Moment of Inertia – Solid – WRT Axis

$$I_{AA\prime} = \sum_{i} r_{i}^{2} \Delta m_{i}$$

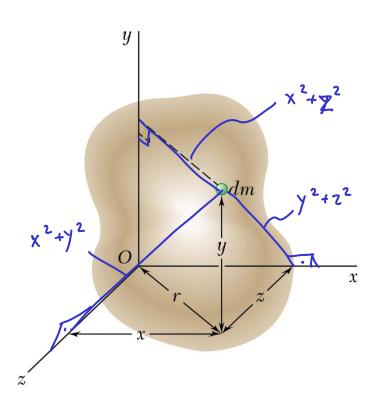






Moment of Inertia – Solid – WRT Coordinate Frame

$$I_{yy} = \int r^2 dm = \int (z^2 + x^2) dm = \iiint_{v} (z^2 + x^2) \rho dv$$
$$I_{xx} = \iiint_{v} (z^2 + y^2) \rho dv$$
$$I_{zz} = \iiint_{v} (x^2 + y^2) \rho dv$$



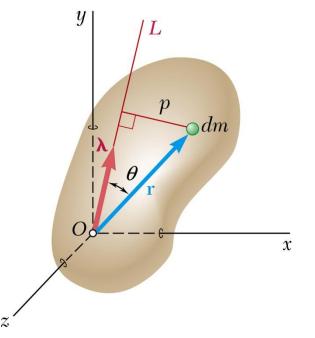




 $p = r \sin\theta = \lambda \times r$

$$I_{\rm OL} = \int p^2 \, dm = \int (\lambda \times r)^2 \, dm = \int (\lambda \times r)^T (\lambda \times r) \, dm$$

$$\lambda \times r = \begin{vmatrix} i & j & k \\ \lambda_x & \lambda_y & \lambda_z \\ x & y & z \end{vmatrix} = i (\lambda_y z - \lambda_z y) + j (\lambda_z x - \lambda_x z) + k (\lambda_x y - \lambda_y x)$$



$$I_{OL} = \int (\lambda_x y - \lambda_y x)^2 + (\lambda_y z - \lambda_z y)^2 + (\lambda_z x - \lambda_x z)^2 dm$$





$$I_{OL} = \int (\lambda_x y - \lambda_y x)^2 + (\lambda_y z - \lambda_z y)^2 + (\lambda_z x - \lambda_x z)^2 dm$$

$$I_{XX} \qquad I_{YY} \qquad I_{ZZ}$$

$$I_{OL} = \lambda_x^2 \int (y^2 + z^2) dm + \lambda_y^2 \int (z^2 + x^2) dm + \lambda_z^2 \int (x^2 + y^2) dm$$

$$-2\lambda_x \lambda_y \int xy dm - 2\lambda_y \lambda_z \int yz dm - 2\lambda_z \lambda_x \int zx dm$$

$$I_{XY} \qquad I_{YZ} \qquad I_{ZX}$$

$$I_{OL} = I_{xx}\lambda_x^2 + I_{yy}\lambda_y^2 + I_{zz}\lambda_z^2 - 2I_{xy}\lambda_x\lambda_y - 2I_{yz}\lambda_y\lambda_z - 2I_{zx}\lambda_z\lambda_z$$





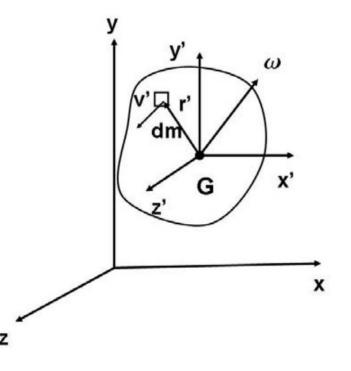
- Expression of the angular momentum of a system of particles about the center of mass, the angular momentum H_G is defined as

$$oldsymbol{H}_G = \sum_{i=1}^n (oldsymbol{r}'_i imes m_i(oldsymbol{\omega} imes oldsymbol{r}'_i)) = \sum_{i=1}^n m_i r_i'^2 oldsymbol{\omega}$$

Where, r' is the position vector relative to the center of mass, v' is the velocity relative to the center of mass.

• For a 3D continuum mass of a rigid body, the summation can be replaced by an integration over the entire mass.

$$oldsymbol{H}_G = \int_m oldsymbol{r}' imes oldsymbol{v}' \ dm$$







• For a 3D rigid body, the distance r' between infinitesimal mass dm and the center of mass G remains constant, and the infinitesimal mass velocity v', relative to the center of mass G, due to the rotation of the rigid body by an angular velocity ω is expressed by

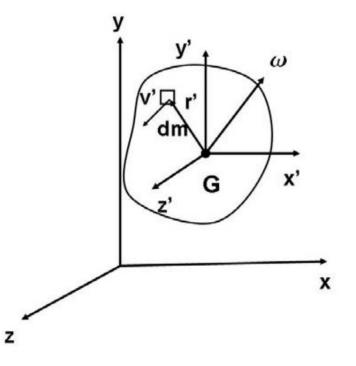
 $oldsymbol{v}' = oldsymbol{\omega} imes oldsymbol{r}'$

• Using the vector identity

$$A \times (B \times C) = (A \cdot C)B - (A \cdot B)C$$

the expression the angular momentum H_G is rewritten as

$$\boldsymbol{H}_{G} = \int_{m} \boldsymbol{r}' \times (\boldsymbol{\omega} \times \boldsymbol{r}') \ dm = \int_{m} [(\boldsymbol{r}' \cdot \boldsymbol{r}') \boldsymbol{\omega} - (\boldsymbol{r}' \cdot \boldsymbol{\omega}) \boldsymbol{r}'] \ dm$$



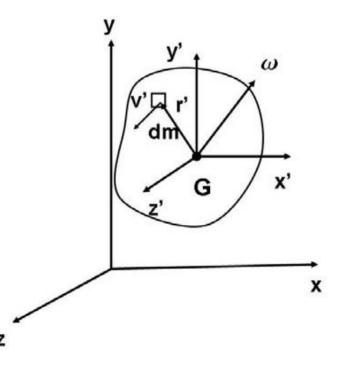




• For a 2D rigid body, rotating in its own plane the distance r'between infinitesimal mass dm is perpendicular to the angular velocity ω , therefore the term $r' \cdot \omega$ is zero, as a result the angular velocity vector ω is parallel to the angular momentum H_G

$$\begin{aligned} \boldsymbol{H}_{G} &= \int_{m} \boldsymbol{r}' \times (\boldsymbol{\omega} \times \boldsymbol{r}') \, dm = \int_{m} [(\boldsymbol{r}' \cdot \boldsymbol{r}') \boldsymbol{\omega} - (\boldsymbol{r}' / \boldsymbol{\omega}) \boldsymbol{r}'] \, dm \\ H_{G} &= \int_{m} (\boldsymbol{r}' \cdot \boldsymbol{r}') \boldsymbol{\omega} \end{aligned}$$

• In the three-dimensional case however, this simplification does not occur, and as a consequence, the angular velocity vector, ω , and the angular momentum vector, H_G , are in general, not parallel.







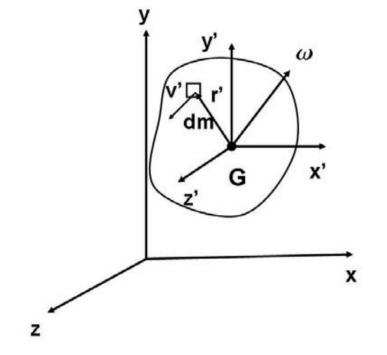
• In cartesian coordinates, the distance r' between infinitesimal mass dm and the center of mass G and the angular velocity vector, ω are defined as

$$egin{aligned} m{r}' &= x'm{i} + y'm{j} + z'm{k} \ m{\omega} &= \omega_xm{i} + \omega_ym{j} + \omega_zm{k} \end{aligned}$$

• The expression for the the angular momentum H_G can be expended to

$$\begin{aligned} \boldsymbol{H}_{G} &= \left(\omega_{x} \int_{m} (x'^{2} + y'^{2} + z'^{2}) \, dm - \int_{m} (\omega_{x}x' + \omega_{y}y' + \omega_{z}z')x' \, dm \right) \boldsymbol{i} \\ &+ \left(\omega_{y} \int_{m} (x'^{2} + y'^{2} + z'^{2}) \, dm - \int_{m} (\omega_{x}x' + \omega_{y}y' + \omega_{z}z')y' \, dm \right) \boldsymbol{j} \\ &+ \left(\omega_{z} \int_{m} (x'^{2} + y'^{2} + z'^{2}) \, dm - \int_{m} (\omega_{x}x' + \omega_{y}y' + \omega_{z}z')z' \, dm \right) \boldsymbol{k} \\ &= \left(I_{xx}\omega_{x} - I_{xy}\omega_{y} - I_{xz}\omega_{z} \right) \boldsymbol{i} \\ &+ \left(-I_{yx}\omega_{x} + I_{yy}\omega_{y} - I_{yz}\omega_{z} \right) \boldsymbol{j} \end{aligned}$$

+ $(-I_{zx}\omega_x - I_{zy}\omega_y + I_{zz}\omega_z) \mathbf{k}$.





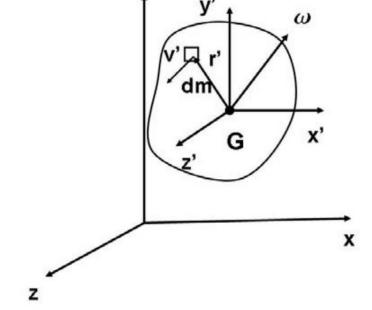


• The quantities I_{xx} , I_{yy} , and I_{zz} are called the mass moments of inertia with respect to the x, y and z axis, respectively, and are given by

$$I_{xx} = \int_{m} (y'^{2} + z'^{2}) dm = \iiint_{V} (y'^{2} + z'^{2}) \rho dv$$

$$I_{yy} = \int_{m} (x'^{2} + z'^{2}) dm = \iiint_{V} (x^{2} + z'^{2}) \rho d$$

$$I_{zz} = \int_{m} (x'^{2} + y'^{2}) dm = \iiint_{V} (x'^{2} + y'^{2}) \rho d$$
Mass moments of inertia



- We observe that the quantity in the integrand is precisely the square of the distance to the x, y and z axis, respectively.
- It is also clear, from their expressions, that the moments of inertia are always positive





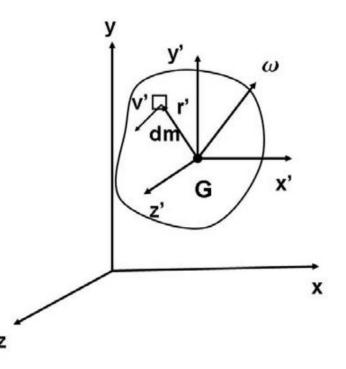
• The quantities I_{xy} , I_{yx} , I_{xz} , I_{zx} , I_{yz} , and I_{zy} are called mass products of inertia and they can be positive, negative, or zero, and are given by,

$$I_{xy} = I_{yx} = \int_{m} (x'y')dm = \iiint_{V} x'y' \rho d$$

$$I_{xz} = I_{zx} = \int_{m} (x'z')dm = \iiint_{V} x'z' \rho d$$

$$I_{yz} = I_{zy} = \int_{m} (y'z')dm = \iiint_{V} y'z' \rho d$$
Mass products of inertia

• They are a measure of the imbalance in the mass distribution.



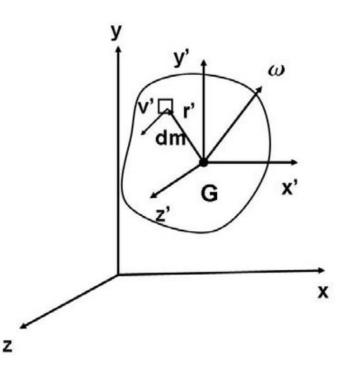




• The angular momentum with respect to the center of mass G can be expressed in a matrix form as

$$oldsymbol{H}_G = [I_G]oldsymbol{\omega}$$

$$\begin{pmatrix} H_{Gx} \\ H_{Gy} \\ H_{Gz} \end{pmatrix} = \begin{pmatrix} I_{xx} & -I_{xy} & -I_{xz} \\ -I_{yx} & I_{yy} & -I_{yz} \\ -I_{zx} & -I_{zy} & I_{zz} \end{pmatrix} \begin{pmatrix} \omega_x \\ \omega_y \\ \omega_z \end{pmatrix}$$



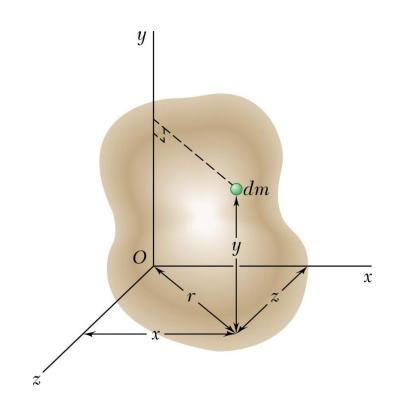




Inertia Tensor – WRT an Arbitrary Coordinate Frame

- For a rigid body that is free to move in a 3D space there are infinite possible rotation axes
- The intertie tensor characterizes the mass distribution of the rigid body with respect to a specific coordinate system
- The intertie Tensor relative to frame {A} is express as a matrix

$${}^{A}I = \begin{bmatrix} I_{xx} & -I_{xy} & -I_{xz} \\ -I_{xy} & I_{yy} & -I_{yz} \\ -I_{xz} & -I_{yz} & I_{zz} \end{bmatrix}$$







Inertia Tensor

$${}^{A}I = \begin{bmatrix} I_{xx} & -I_{xy} & -I_{xz} \\ -I_{xy} & I_{yy} & -I_{yz} \\ -I_{xz} & -I_{yz} & I_{zz} \end{bmatrix}$$

$$I_{xx} = \iiint_{V} (y^{2} + z^{2}) \rho dv$$

$$I_{yy} = \iiint_{V} (x^{2} + z^{2}) \rho d$$

$$I_{zz} = \iiint_{V} (x^{2} + y^{2}) \rho d$$

$$I_{zz} = \iiint_{V} (x^{2} + y^{2}) \rho d$$

$$I_{zz} = \iiint_{V} (x^{2} + y^{2}) \rho d$$

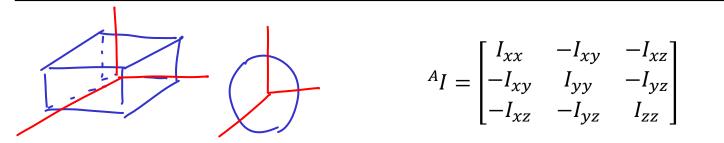
$$I_{zz} = \iiint_{V} (yz \rho d)$$

$$I_{yz} = \iiint_{V} yz \rho d$$

$$I_{yz} = \iiint_{V} yz \rho d$$



Tensor of Inertia – Example



- This set of six independent quantities for a given body, depend on the <u>position and orientation</u> of the frame in which they are defined
- We are free to choose the orientation of the reference frame. It is possible to cause the product of inertia to be zero

$$I_{xy} = 0 I_{xz} = 0 I_{yz} = 0$$
 Mass products of inertia
$$^{A}I = \begin{bmatrix} I_{xx} & 0 & 0 \\ 0 & I_{yy} & 0 \\ 0 & 0 & I_{zz} \end{bmatrix}$$

 The axes of the reference frame when so aligned are called the <u>principle axes</u> and the corresponding mass moments are called the principle <u>moments of intertie</u>





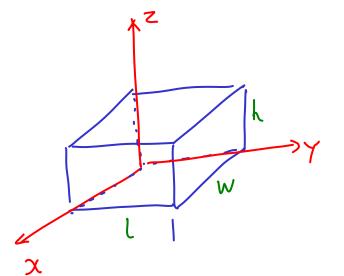
Tensor of Inertia – Example

 $x: 0 \to w$

 $y: 0 \rightarrow l$

 $z: 0 \rightarrow h$

$$I_{xx} = \int_{0}^{h} \int_{0}^{l} \int_{0}^{w} (y^{2} + z^{2}) \rho dx dy dz = \int_{0}^{h} \int_{0}^{l} (y^{2} + z^{2}) w \rho dy dz$$
$$= \left(\frac{hl^{3}w}{3} + \frac{h^{3}lw}{3}\right)\rho = \rho hlw \frac{l^{2}}{3} phlw \frac{h^{2}}{3} = \int_{0}^{h} \left(\frac{l^{3}}{3} + z^{2}l\right) w \rho dz = \frac{m}{3}(l^{2} + h^{2})$$



$$I_{yy} = \frac{m}{3}(w^2 + h^2)$$
$$I_{zz} = \frac{m}{3}(l^2 + w^2)$$





Tensor of Inertia – Example

$$I_{xy} = \int_{0}^{h} \int_{0}^{l} \int_{0}^{w} xy \rho dx dy dz = \int_{0}^{h} \int_{0}^{l} \frac{w^{2}}{2} y \rho dy dz = \int_{0}^{h} \frac{w^{2} l^{2}}{4} \rho dz = \frac{w^{2} l^{2} h}{4} \rho = (w l h \rho) \frac{w l}{4} = \frac{m}{4} w l$$

$$I_{xz} = \frac{m}{4} h w$$

$$I_{yz} = \frac{m}{4} h l$$

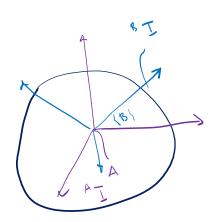
$$A_{I} = \begin{bmatrix} \frac{m}{3} (l^{2} + h^{2}) & -\frac{m}{4} w l & -\frac{m}{4} h w \\ -\frac{m}{4} w l & \frac{m}{2} (w^{2} + h^{2}) & -\frac{m}{4} h l \end{bmatrix}$$

 $\begin{bmatrix} -\frac{1}{4}wl & \frac{1}{3}(w^{2}+h^{2}) & -\frac{1}{4}hl \\ -\frac{1}{4}hw & -\frac{1}{4}hl & \frac{1}{3}(l^{2}+w^{2}) \end{bmatrix}$

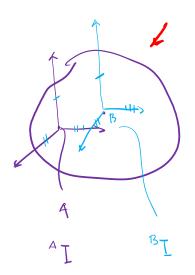




Tensor of Inertia – Operations



Translations of the Inertia Tensor Parallel Axis Theorem





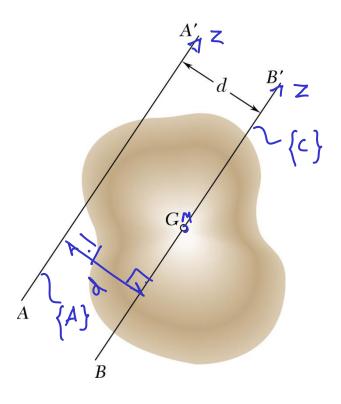


- The inertia tensor is a function of the position and orientation of the reference frame
- **Parallel Axis Theorem** How the inertia tensor changes under translation of the reference coordinate system

Frame $\{C\}$ – is located at the CM

Frame {A} – an arbitrarily translated frame

$$^{A}I_{zz} = {}^{C}I_{zz} + md^{2}$$



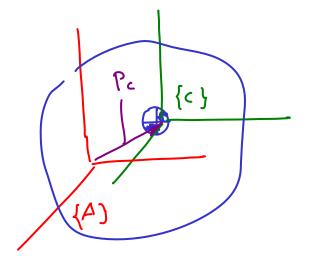




$${}^{A}I_{xx} = {}^{C}I_{xx} + m(z_{c}^{2} + y_{c}^{2})$$
$${}^{A}I_{yy} = {}^{C}I_{yy} + m(x_{c}^{2} + z_{c}^{2})$$
$${}^{A}I_{zz} = {}^{C}I_{zz} + m(x_{c}^{2} + y_{c}^{2})$$

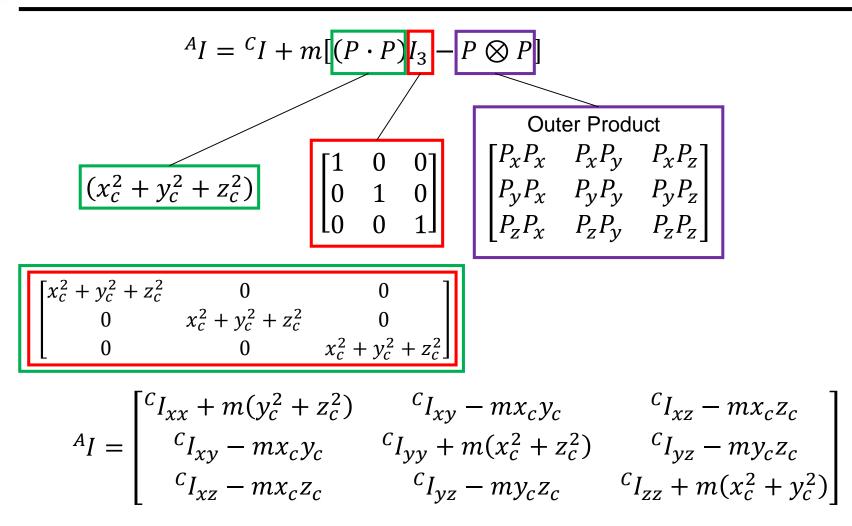
$${}^{A}I_{xy} = {}^{C}I_{xy} - mx_{c}y_{c}$$
$${}^{A}I_{xz} = {}^{C}I_{xz} - mx_{c}z_{c}$$
$${}^{A}I_{yz} = {}^{C}I_{yz} - my_{c}z_{c}$$

$$P_c = \begin{bmatrix} x_c \\ y_c \\ z_c \end{bmatrix}$$
 - Location of the CM (origin of C) relative to frame [A]













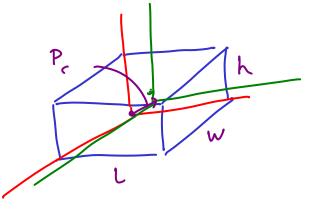
Tensor of Inertia – Example

$$P_{c} = \begin{bmatrix} x_{c} \\ y_{c} \\ z_{c} \end{bmatrix} = \frac{1}{2} \begin{bmatrix} w \\ l \\ h \end{bmatrix}$$

$$^{c}I_{zz} = ^{A}I_{zz} - m(x_{c}^{2} + y_{c}^{2}) = \frac{m}{3}(l^{2} + w^{2}) - \frac{m}{4}(w^{2} + l^{2}) = \frac{m}{12}(w^{2} + l^{2})$$

$$^{c}I_{xy} = ^{A}I_{xy} + mx_{c}y_{c} = -\frac{mwl}{4} + m\frac{1}{2}w\frac{1}{2}l = 0$$

$$^{c}I = \frac{m}{12} \begin{bmatrix} h^{2} + l^{2} & 0 & 0 \\ 0 & w^{2} + h^{2} & 0 \\ 0 & 0 & l^{2} + w^{2} \end{bmatrix}$$







Tensor of Inertia – Operations

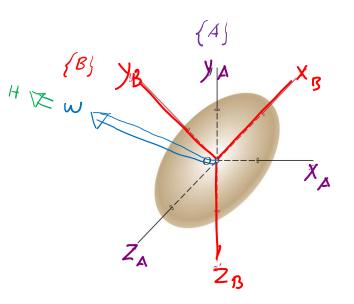
Rotation of the Inertia Tensor





Rotation of the Inertia Tensor

- Given:
 - The inertia tensor of the a body expressed in frame A
 - Frame B is rotated with respect to frame A
 - Note: Both frames are stationary in space
- Calculate
 - The inertia tensor of the body expressed in frame B







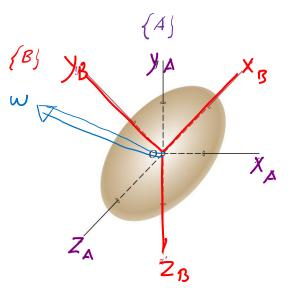
Rotation of the Inertia Tensor

$${}^{A}H = {}^{A}I {}^{A}\omega$$
$${}^{B}H = {}^{B}I {}^{B}\omega$$

 $(*)^A \omega$, ^{*A*}*H* - angular velocity and momentum expressed in frame A

 $(*)^{B}\omega$, ^BH - angular velocity and momentum expressed in frame B

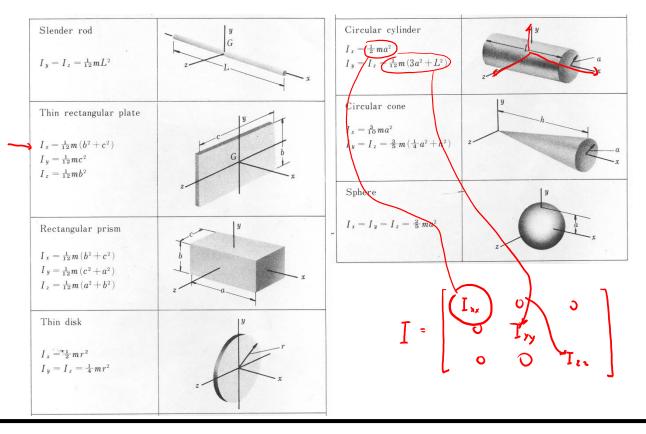
$$AH = {}^{A}_{B}R^{B}H$$
$$A\omega = {}^{A}_{B}R^{B}\omega$$
$$AH = {}^{A}_{B}R^{B}I^{B}\omega$$
$$AH = {}^{A}_{B}R^{B}I({}^{A}_{B}R^{-1}{}^{A}_{B}R)^{B}\omega$$
$$AH = {}^{A}_{B}R^{B}I^{A}_{B}R^{-1}A\omega$$
$$I_{A} = {}^{A}_{B}R^{B}I^{A}_{B}R^{-1} = {}^{A}_{B}R^{B}I^{A}_{B}R^{T}$$







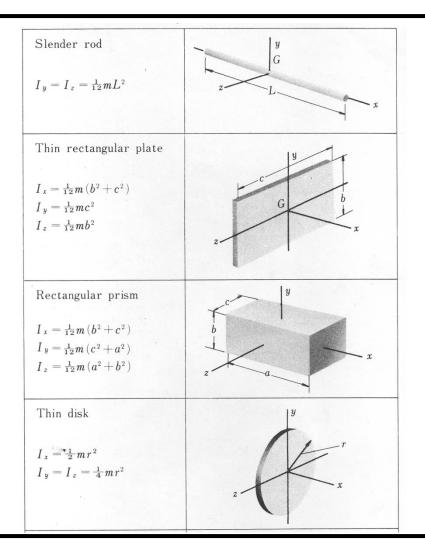
• The elements for relatively simple shapes can be solved from the equations describing the shape of the links and their density. However, most robot arms are far from simple shapes and as a result, these terms are simply measured in practice.







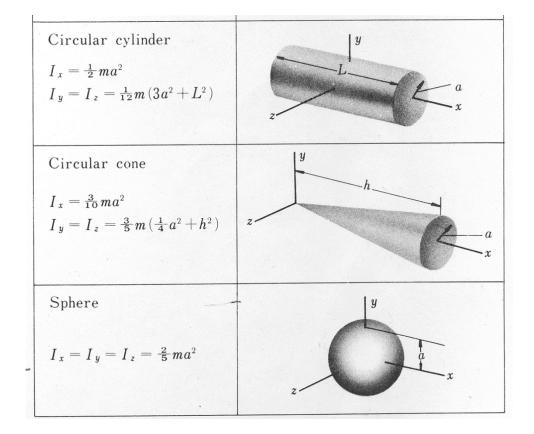
Inertia Tensor 2/





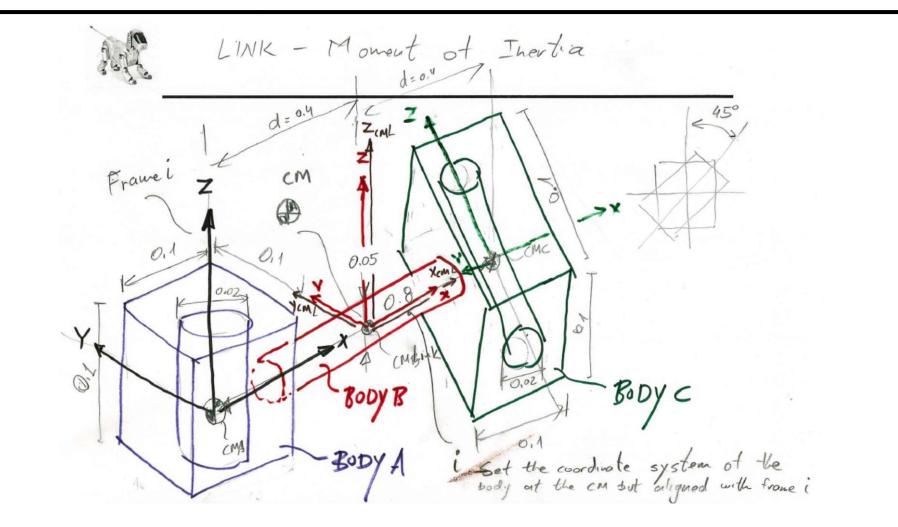


Inertia Tensor 2/



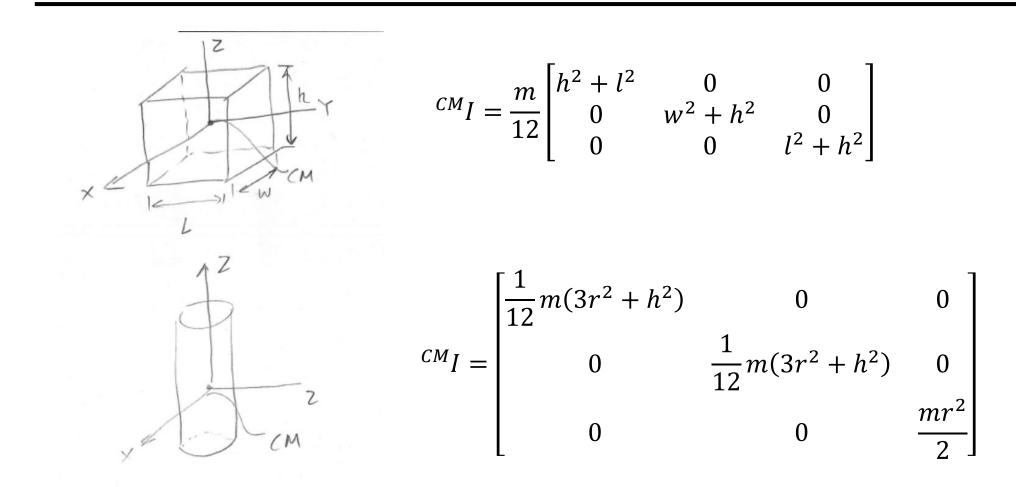








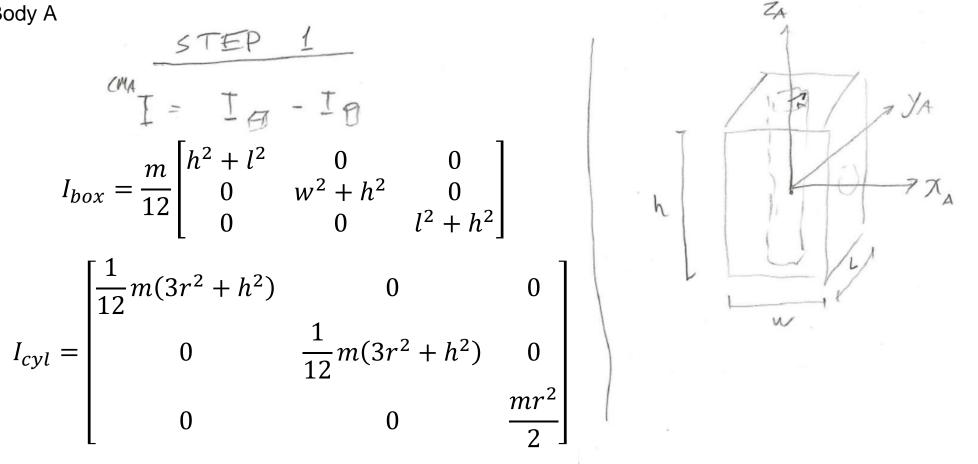








Body A ٠



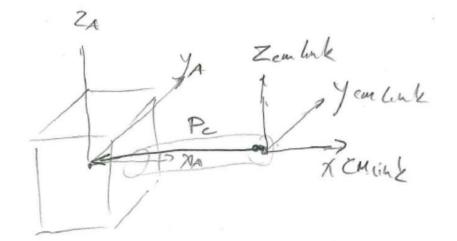




• Body A

STEP 2 – Translate from frame A to the frame at the CoM of the link

$$^{CM,link}I = {}^{CM,A}I + m[P_c^T P_c I_3 - P \otimes P]$$

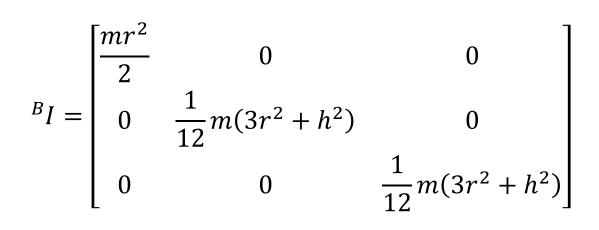


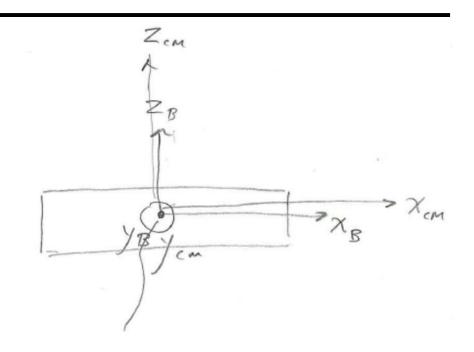
$$= {}^{CM,A}I + m \left[\begin{bmatrix} -d & 0 & 0 \end{bmatrix} \begin{bmatrix} -d \\ 0 \\ 0 \end{bmatrix} I_3 - \begin{bmatrix} d^2 & 0 & 0 \\ 0 & 0 & 0 \\ 0 & 0 & 0 \end{bmatrix} \right]$$





• Body B



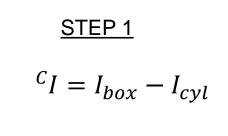


The frame of body B is aligned with the frame of the CM of the entire body

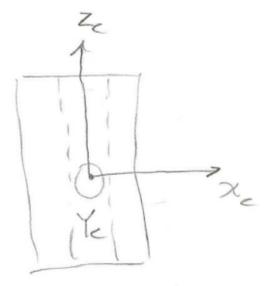




• Body C



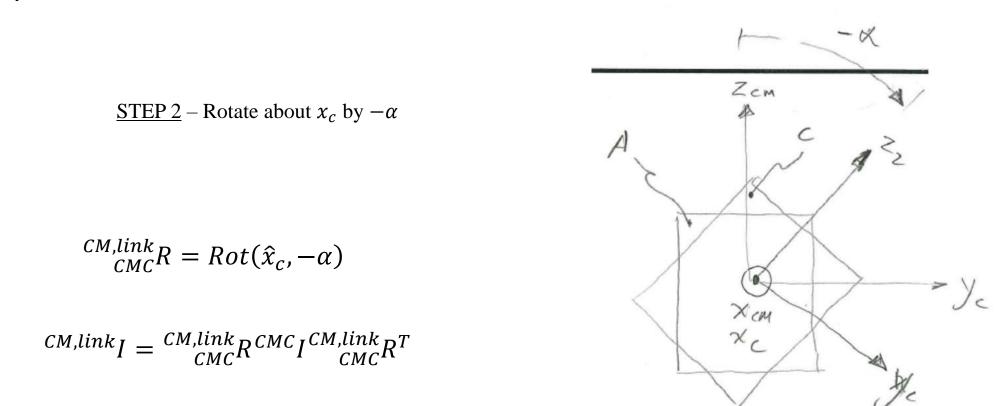
See body A





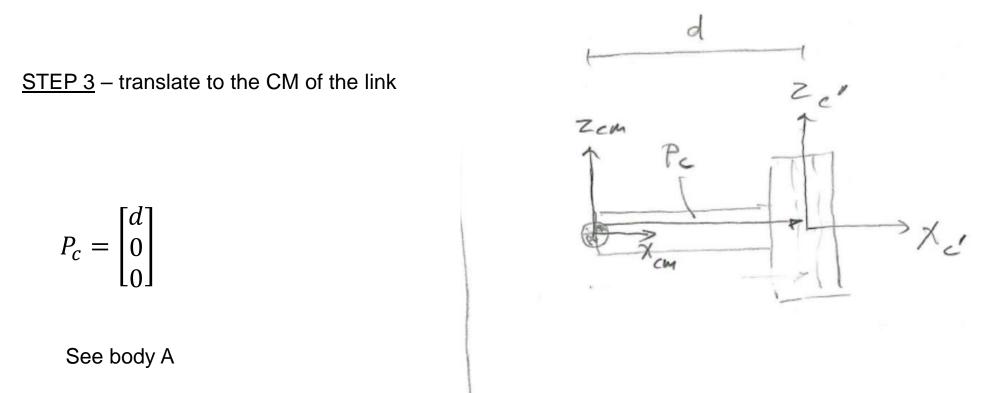


• Body C





• Body C







Summary





 I_{yy}

The angular momentum with respect to the center of mass G can ٠ be expressed in a matrix form as

$$H_{G} = [I_{G}]\omega$$

$$\begin{pmatrix} H_{Gx} \\ H_{Gy} \\ H_{Gz} \end{pmatrix} = \begin{pmatrix} I_{xx} & -I_{xy} & -I_{xz} \\ -I_{yx} & I_{yy} & -I_{yz} \\ -I_{zx} & -I_{zy} & I_{zz} \end{pmatrix} \begin{pmatrix} \omega_{x} \\ \omega_{y} \\ \omega_{z} \end{pmatrix}$$

$$I_{xx} = \int_{m} (y'^{2} + z'^{2}) dm = \iiint_{V} (y'^{2} + z'^{2}) \rho dv$$

$$I_{xy} = I_{yx} = \int_{m} (x'y') dm = \iiint_{V} x'y' \rho d$$

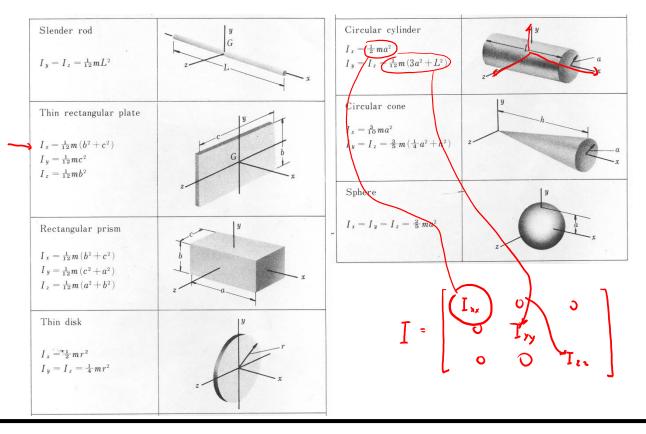
$$I_{xz} = I_{zx} = \int_{m} (x'z' + y'^{2}) dm = \iiint_{V} (x'^{2} + y'^{2}) \rho d$$

$$I_{yz} = I_{zy} = \int_{m} (y'z') dm = \iiint_{V} y'z' \rho d$$





• The elements for relatively simple shapes can be solved from the equations describing the shape of the links and their density. However, most robot arms are far from simple shapes and as a result, these terms are simply measured in practice.



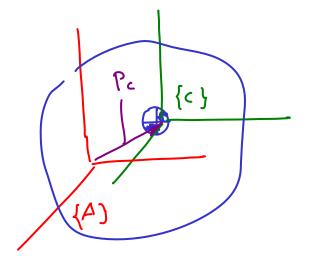




$${}^{A}I_{xx} = {}^{C}I_{xx} + m(z_{c}^{2} + y_{c}^{2})$$
$${}^{A}I_{yy} = {}^{C}I_{yy} + m(x_{c}^{2} + z_{c}^{2})$$
$${}^{A}I_{zz} = {}^{C}I_{zz} + m(x_{c}^{2} + y_{c}^{2})$$

$${}^{A}I_{xy} = {}^{C}I_{xy} - mx_{c}y_{c}$$
$${}^{A}I_{xz} = {}^{C}I_{xz} - mx_{c}z_{c}$$
$${}^{A}I_{yz} = {}^{C}I_{yz} - my_{c}z_{c}$$

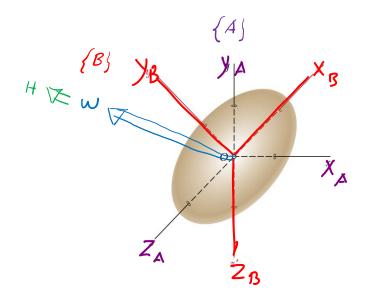
$$P_c = \begin{bmatrix} x_c \\ y_c \\ z_c \end{bmatrix}$$
 - Location of the CM (origin of C) relative to frame [A]







Rotation of the Inertia Tensor



$$I_A = {}^A_B R^B I^A_B R^T$$